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Numerical analysis of the combustion process in a four-stroke compressed natural gas engine with direct injection system[†]

Wendy Hardyono Kurniawan and Shahrir Abdullah*

Department of Mechanical and Materials Engineering, National University of Malaysia, 43600 UKM Bangi, Selangor Darul Ehsan, Malaysia

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Abstract

In this paper, a numerical study to simulate and analyze the combustion process occurred in a compressed natural gas direct injection (CNG-DI) engine by using a multi-dimensional computational fluid dynamics (CFD) code was presented. The investigation was performed on a single cylinder of the 1.6-liter engine running at wide open throttle at a fixed speed of 2000 rpm. The mesh generation was established via an embedded algorithm for moving meshes and boundaries for providing a more accurate transient condition of the operating engine. The combustion process was characterized with the eddy-break-up model of Magnussen for unpremixed or diffusion reaction. The modeling of gaseous fuel injection was described to define the start and end of injection timing. The utilized ignition strategy into the computational mesh was also explained to obtain the real spark ignition timing. The natural gas employed is considered to be 100% methane (CH₄) with three global step reaction scheme. The CFD simulation was started from the intake valves opening until the time before exhaust valves opening. The results of CFD simulation were then compared with the data obtained from the single-cylinder engine experiment and showed a close agreement. For verification purpose, comparison between numerical and experimental work are in the form of average in-cylinder pressure, engine power as well as emission level of CO and NO.

Keywords: Combustion; Computational fluid dynamics; Compressed natural gas; Direct injection; Diffusion reaction; Ignition

1. Introduction

With the great attention to energy availability today, much effort has been directed to development for substituting carbon-based oil fuels. Utilization natural gas is seen as one of the promising solutions to resolve this kind of problem. While fuel cell is considered as a power source in the future, the use of natural gas for automotive vehicles in their internal combustion is more practicable and cheaper. Several advantages related to natural gas utilization in terms of engineering application are its higher thermal efficiency and lower exhaust emissions including CO_2 due to the higher octane level and lower ratio of carbon and hydrogen ratio, respectively.

For light-duty vehicle, direct injection (DI) of compressed natural gas (CNG) promises higher thermal efficiencies comparable to those achieved by high compression ratio and unthrottled diesel engines, while maintaining the smoke-free operation of spark ignition (SI) engines and producing slightly higher NOx emissions with the proper operating conditions. In order to meet the more stringent emission regulations that has been implemented all around the world, there are still technical difficulties to be resolved to simultaneously reduce both NO and CO produced from the natural gas engines.

The CFD modeling of internal combustion engine (ICE) process has one the highest level of complexity and it is a challenging task given the fact that several

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 ^{*}Corresponding author. Tel.: +60 3 8921 6013, Fax.: +60 3 8925 9659 E-mail address: shahrir@eng.ukm.my

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processes, parameters and operating conditions occurred within the engine should be taken into account such as fuel injection, flame propagation, ignition process, chemical kinetic reaction, exhaust emission formations, tendency for knocking as well as control of air-fuel ratio control. Hence, the computing times of ICE process simulation are costly and require huge computer memory as well as high performance computing setup.

A number of automotive researchers have initiated works on natural gas-fuelled engines by performing experiments or numerical simulation. The characteristics of combustion and exhaust gas emissions from a dual-fuel diesel engine with natural gas by the means of experimental work and numerical simulation have been performed by Kusaka et al. [1], where numerical computation was performed by using KIVA-3 to analyze the formation of NO_x and hydrocarbon total. It was found that NO_x within cylinder of dual fuel engine is higher than that of diesel engine due to higher temperature at the initial stage of combustion process. Shiga et al. [2] have also analyzed emissions characteristics from compressed natural gas (CNG) direct-injection combustion using a rapid-compression-machine with a compression ratio of 10:1 and a disc-shaped combustion chamber. In addition, Ando et al. [3] conducted their experimental with a selfignited natural gas engine by performing the visualization and combustion study. In numerical analysis, Zhang and Frankel [4] carried out combustion analysis of natural gas engine by using the CFD code under the lean combustion and limited analyzed CFD calculation. Comparison of fuel composition and ignition energy at the initial stage of combustion inside a natural gas SI engine was studied by Yossefi et al. [5] by using KIVA-2 and the detailed chemical kinetic algorithm to compare the relatively effect of gas composition, i.e. ethane and carbon dioxide inside natural gas. Agarwal and Assanis [6] carried out detailed multidimensional modeling of a direct injection natural gas engine with self-ignited condition for the purpose of examining ignition, combustion and formation processes of NO by using KIVA-3 with the eddy-break-up model, coupled with chemical kinetic mechanism for natural gas simulation. Lastly, Zheng et al. [7] investigated fluid flow and combustion process using CFD in a compression ignition natural gas engine with separated chamber.

From literature, numerical analysis of a natural gas engine with spark ignition mounted inside the combustion chamber has yet to be covered in details. Therefore, numerical study of combustion process in a CNG engine with direct injection system, referred to as CNGDI, was carried out. The simulations for the engine operating conditions at the moderate engine speed of 2000 rpm were carried out to investigate the behavior occurred within the engine cylinder during and after combustion process.

2. Engine configuration

A single cylinder engine was modified into a CNGDI monofuel engine. The engine was operated at wide open throttle condition with a compression ratio of 14:1. The main engine specifications are as given in Table 1 and the section view of its cylinder head and liner is depicted in Fig. 1, which shows the position of the intake and exhaust ports, intake and exhaust valves, CNG injector, spark plug and combustion chamber with the shape of piston bowl.

The CFD simulation for the combustion process in this analysis were performed by using the moving mesh and boundary algorithm where every event represents the different mesh and boundary geometries for every different crank angle in each step of engine cycle. Thus, in order to perform CFD simulation for internal combustion process, the analysis was carried out by using transient, moving meshes and boundaries, high compressible Reynolds number, high turbulence intensity, momentum, heat and mass transfer with complex geometries model and chemical-thermal dependent as well [8].

A self-developed grid generation program had been

Table 1. Specification of CNGDI engine. Value Unit Engine parameters Number of cylinders 4 Type Inline Displacement volume 1596 cm Bore 78 mm Stroke 84 mm Connecting rod length 131 mm Crank radius 44 mm 14 Compression ratio Intake valve opening 12 bTDC Intake valve closing 48 aBDC Exhaust valve opening 45 bBDC Exhaust valve closing 10 aTDC Maximum intake valve lift 8.1 mm

7.5

mm

Maximum exhaust valve lift

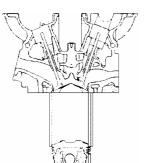


Fig. 1. The geometrical details of CNG-DI engine.

employed to produce hexahedral cells for CFD computational mesh of the engine model, which involves the intake ports and valves, the cylinder head and the piston bowl as shown in Fig. 2. The number of cells was about 43,398 at top dead centre (TDC) position and around 163,110 cells at bottom dead centre (BDC) position, where about the half of the cells used to generate the mesh at the cylinder head and piston bowl in the case of considering the grid sensitivity and reasonable computer run time. The positions of engine computational model at TDC and BDC can be seen in Fig. 3(a) and 3(b), respectively.

The grid dependency tests had been performed to examine suitability of normal mesh configuration against finer mesh structure and it was found that the current grid construction was sufficient to achieve convergence, accuracy, stability and consistency in the terms of cylinder pressure and temperature. The fine grid arrangement was necessary for the valve movement to obtain the stability and convergence criteria. Hexahedral cells had been employed for mesh generation due to the better accuracy and stability compared to the tetrahedral cells. Grid dependency study had been carried out to determine the normal mesh configuration against the finer mesh. It was found that the current employed grid construction was relatively enough to achieve the convergence, accuracy, stability and consistency in the terms of cylinder pressure, temperature and various species concentrations. The other important consideration of choosing the mesh construction in this paper is computational time. Therefore, an appropriate mesh configuration should be selected and determined for simulation of combustion process with reasonable computational cost. In this work, the required CPU time to simulate the combustion process at the engine speed of 2000 rpm is around 69 hour on the 4-CPU SGI Origin 300 clustered computer.

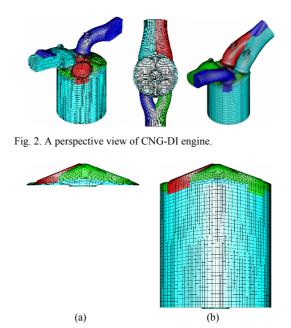


Fig. 3. Engine model at (a) TDC and (b) BDC.

3. Methodology

The numerical simulation was performed using Star-CD equipped with the moving mesh and boundary capability for characterizing real engine operating conditions, such as valves and piston movement. In this section, the chemistry model and the reaction mechanism utilized for CFD simulation of the combustion process were described together with initial and boundary conditions.

The set of governing equations which were employed to simulate the combustion process mentioned above consists of mass, momentum and energy equations as well as the state equation for ideal gas. Those equations used, which are referred to as the general governing equations in CFD analysis, are described in vector notation as:

Continuity equation:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0 \tag{1}$$

Momentum equation:

$$\rho \frac{\partial \mathbf{u}}{\partial t} + \rho(\mathbf{u} \cdot \nabla)\mathbf{u} = -\nabla p +$$

$$\nabla \cdot \left\{ \mu \left[\nabla \mathbf{u} + (\nabla \mathbf{u})^{\mathrm{T}} \right] + \lambda (\nabla \cdot \mathbf{u}) \right\} + \rho \mathbf{g}$$
(2)

1940

Energy equation:

$$\rho c_{\mathcal{V}} \frac{\partial T}{\partial t} + \rho c_{\mathcal{V}} (\mathbf{u} \cdot \nabla) T = -p(\nabla \cdot \mathbf{u}) + \nabla (k \nabla T) +$$

$$\lambda (\nabla \cdot \mathbf{u})^{2} + \nabla \mathbf{u} \cdot \left\{ \mu \left[\nabla \mathbf{u} + (\nabla \mathbf{u})^{T} \right] \right\}$$
(3)

The turbulence model employed was the k- ε turbulence model [9] for compressible and high Reynolds number flow for reciprocating engines, where k is turbulence kinetic energy and ε is turbulence dissipation rate. As mentioned previously, the finite volume method CFD code employed to solve the discretized equations, which govern the mean fluid motion, is based on the pressure-correction method. The discretization of space and time are maintained and monitored based on a specified Courant number, which provide stability for time discretization. The time step was defined to obtain the best compromise between requirement for convergence and computing load i.e. smaller time steps are used for initiating simulations and when reaching fuel injection event and TDC position

The temporal discretization method chosen was implicit with the under relaxation factor of 0.1 to obtain criterion for unconditional numerical stability, while the accurate second order differencing scheme of MARS (monotone advection and reconstruction scheme) was employed for the momentum, energy and turbulence equations together with the arbitrary Lagrangian-Eulerian (ALE) technique to treat grid movement associated with movement of piston and valves. The popular PISO algorithm for unsteady flows was then employed to solve the resulting algebraic equations.

The type of chemical reaction model used to simulate the combustion process of compressed natural gas was unpremixed or diffusion reaction. The motivation of using the diffusion reaction for such engine modeling is due to the monofuel employed (methane). In this reaction, the fuel and the oxidant (air) enter the engine cylinder separately. For modeling, it is necessary to solve one additional differential conservation equation for each reaction considered. The chosen conserved scalar is known as the mixture fraction, ϕ , and is defined as the total mass fraction of burned and unburned fuel. Hence, ϕ is obtained by solving the following equation:

$$\frac{\partial}{\partial t}(\rho\phi) + \nabla \cdot (\rho \mathbf{u}\phi) = \nabla \cdot \left(\rho\Gamma + \frac{\mu_t}{\sigma_t}\right) \frac{\partial\phi}{\partial x_j} + S \qquad (4)$$

For the simulation of the turbulent combustion, the eddy break-up (EBU) model of Magnussen [10] was implemented in the present work. The EBU model was usually developed and utilized for any combustion applications and was based on the two assumptions; that as the reaction is single-step irreversible involving fuel (F), oxidant (O) and products (P), plus possible background inert species, and that the reaction time scale is so small such that the rate-controlling mechanism is turbulent micro-mixing. According to the employed model, the fuel consumption rate is derived as follows:

$$R_{\rm F} = -\frac{\rho\varepsilon}{k} A_{\rm EBU} \cdot \min\left[Y_{\rm F}, \frac{Y_{\rm O}}{s_{\rm O}}, B_{\rm EBU} \frac{Y_{\rm P}}{s_{\rm P}}\right] \text{kg/m}^3 \text{s}$$

$$s_{\rm O} \equiv \frac{n_{\rm O}M_{\rm O}}{n_{\rm F}M_{\rm F}} \text{ and } s_{\rm P} \equiv \frac{n_{\rm P}M_{\rm P}}{n_{\rm F}M_{\rm F}}$$
(6)

where A_{EBU} and B_{EBU} are dimensionless empirical coefficients with nominal modified values for higher RON, i.e. 12 and 0.8, respectively, according to the burned mass fraction of fuel during combustion. The first two arguments in the square brackets of Eq. (5) determine the local rate-controlling mass fraction, while the third, which may be omitted optionally, is intended to inhibit reaction at low temperature. The micro-mixing time scale is taken to be k/ε , which is the dissipation time scale.

The combustion modeling strategy implemented was based on the three-step global reaction of the EBU model. For natural gas, which is considered as 100% methane (CH₄):

$$CH_4 + 0.5O_2 \rightarrow CO + 2H_2$$

$$CO + 0.5O_2 \rightarrow CO_2$$

$$H_2 + 0.5O_2 \rightarrow H_2O$$
(7)

The mixture was assumed to follow the ideal gas law. Viscosity, thermal conductivity and specific heat of the mixture are computed from the properties of individual species and are all functions of temperature. The mass fractions of the combustion products were assumed to follow local and instantaneous thermody

	1	
Operating parameters	Unit	Value
Start of injection timing	° before TDC	130
End of injection timing	° before TDC	80
Ignition timing	° before TDC	18
Initial pressure	bar	1.04
Initial temperature	К	302
Intake port temperature	К	305
Exhaust port temperature	K	802
Equivalence ratio	-	1.0
Fuel flow rate	g/s	0.505

Table 2. The operating condition at speed of 2000 rpm.

namic equilibrium values. The equilibrium composition of the cylinder charge depends on the pressure, temperature and equivalence ratio. The considered chemical species were CH_4 , O_2 , CO_2 , H_2O , N_2 , H_2 , CO and NO.

The engine operating conditions chosen for the CFD simulation for verification against the experimental data was carried out for an engine speed of 2000 rpm. The engine operating conditions covered certain variations in the intake temperature, injection timing, injection duration and spark ignition timing. The operating conditions taken from SCRE were listed in Table 2. The simulation was executed by defining the events for engine cycle and it is started from 348° crank angle (CA). It finished at 855°CA, where the exhaust valves were opened. The measured intake and exhaust temperature from the experiment has been used as boundary condition.

The initial pressure was positioned at 100 kPa. The initial temperature of 302 K was set up for the engine computational mesh. The time step used for every degree of crank angles for three mentioned processes is 0.1, which means that 10 time steps were needed to calculate one degree of crank angle. The reason of choosing the smaller time steps was to avoid the negative densities occurred during the calculation, especially when the mesh distort during the intake and exhaust opens and closes, respectively. The gas was assumed to be fresh air. The initial value of the turbulent kinetic energy k was assumed to be spatially uniform and was set equal to 3% of the kinetic energy of the mean piston speed.

4. Results and discussions

The numerical result of the CFD simulation is presented in this section. The main objective is to inves-

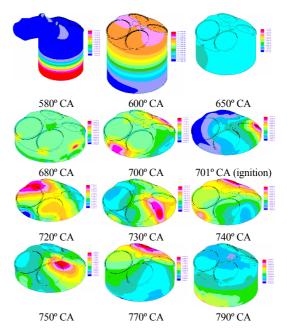


Fig. 4. In-cylinder pressure distribution.

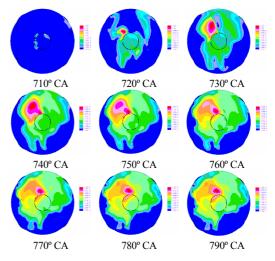


Fig. 5. CO emission formation.

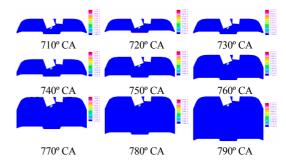


Fig. 6. NO emission formation.

tigate and determine phenomena occurring in the mixture formation and combustion process, such as spray formation during fuel injection, in-cylinder airfuel mixing pressure distribution and formation of CO and NO.

4.1 In-cylinder pressure distribution

The plots of in-cylinder pressure during the combustion process for several crank angles (CA) during compression and power stroke can be depicted in Fig. 9. As predicted, the pressure reached its maximum value at the top part of the cylinder head when the piston reached TDC position during compression stroke, which tend to appear in the side part of engine cylinder, which is near to the spark plug. This shows that the burned fuel causes the flame to propagate to the side part of engine cylinder near the spark plug instead of being distributed uniformly.

4.2 CO emission formation

Fig. 10 presents the level concentration of CO for several crank angles position during and after the combustion. As seen, most of the CO concentration was located next to the cylinder liner wall and inside the piston bowl due to poor combustion, due to oxidation process from CO to CO_2 in the core gas and to wall quenching.

4.3 NO emission formation

Fig. 6 shows the formation of NO emission inside the cylinder. More NO formed around the spark plug due to higher temperature produced during ignition. Apart from high temperature, the amount of NO generated depends on pressure, air-fuel ratio and combustion time within the cylinder. The highest concentration was found around the spark plug, where the highest temperature was observed.

5. Experiment and numerical verification

The experimental setup for the purpose of verifying the CFD simulation was carried out using a SCRE test rig as shown in Fig. 7 (a). During the experiment, all engine boundary conditions were fixed. Pressure sensor (Kistler 6061B) were installed to measure the cylinder pressure occurred during combustion process. To obtain the boundary conditions at the intake and outflow boundaries, thermocouple sensor were in

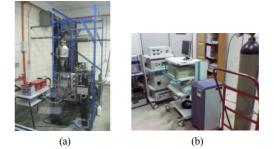


Fig. 7. Experimental setup: (a) SCRE, (b) gas analyzer.

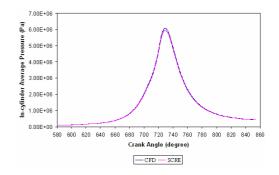


Fig. 8. The calculated and measured in-cylinder pressure.

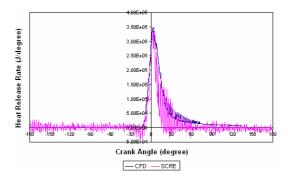


Fig. 9. The calculated and measured heat release rate.

stalled at nearest possible to the cylinder head so that the temperature of intake and exhaust ports can be determined. Fuel injection supply and ignition timing was controlled by a programmable electronic control unit (ECU). An exhaust gas analyzer to measure the exhaust emissions level was used as shown in Fig. 7 (b) and was located at 3 m from the exhaust port.

The simulated and experimental results were compared at 2000 rpm. Fig. 8 illustrates the comparison for in-cylinder pressure curves, while Fig. 9 shows comparison for the heat release rate. Fig. 10 shows the generated P-V diagram with its indicated work is 325 J. Thus, the indicated power resulted from the conversion of indicated work at this speed is 5.42 kW.

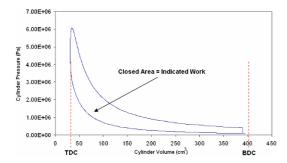


Fig. 10. The P-V diagram for indicated power.

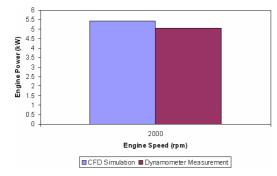


Fig. 11. The simulated and measured engine power.

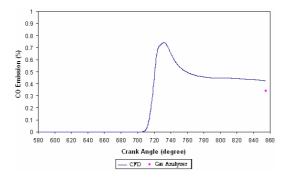


Fig. 12. The calculated and measured CO emission.

As can be seen from Fig. 8, the cylinder pressure obtained from simulation has the higher value due to the fact that this simplified CFD simulation does not include friction losses due interaction of engine components. Hence, having a CFD result slightly higher than the experimental data for verification is acceptable. Then, the heat release rate obtained from simulation was compared with the experimental data and shows a good agreement as given in Fig. 9. Similarly, the comparison of engine power between the CFD simulation and the experimental data by the dynamometer attached to at the SCRE test rig also shows a good agreement.

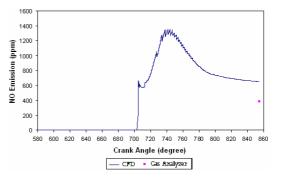


Fig. 13. The calculated and measured NO emission.

Then, the comparison between the engine power achieved from the CFD simulation and experimental data were compared and demonstrated in Fig. 11. On the other hand, the assessment of emissions levels can be shown in Figs. 12 and 13 for both CO and NO values, respectively. These simulated emissions value gives the higher values than the measured levels since the data was measured at the distance of 3 m from the exhaust port.

6. Conclusion

In the last, the modeling work presented has proved that the adopted approach of the combustion process via CFD can be implemented to investigate the other phenomena within the engine cylinder, such as the flame propagation, optimization of injection process and even optimization of combustion parameters. For future work, analysis of combustion processes under different engine speeds and operating conditions will be carried out as well as the optimization of critical engine parameters.

Nomenclature-

- A : Surface area of injection boundary (m^2)
- **g** : Gravitational acceleration (m/s^2)
- k : Thermal conductivity coefficient (W/m·K)
- k : Turbulent kinetic energy (m²/s²)
- p : Pressure (N/m²)
- S : Source term
- t : Time (s)
- T : Temperature (K)
- **u** : Velocity vector (m/s)

Greek Symbols

 ρ : Density (kg/m³)

- μ : Dynamic viscosity (Pa·s)
- c_v : Heat capacity at constant volume (J/kg·K)
- Γ : Diffusivity coefficient
- σ_t : Turbulent Schmidt number
- ϕ : Mixture fraction
- ε : Turbulent dissipation rate (m²/s³)

Subscripts and Superscripts

- F : Fuel
- O : Oxidant
- P : Product

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